

AD647603

Technical

Report

F-B2294

DDC
RECEIVED
MAR 3 1967
B

ON THE EIGENVALUES OF COUETTE
FLOW IN A FULLY-FILLED
CYLINDRICAL CONTAINER

ARCHIVE COPY



THE FRANKLIN INSTITUTE RESEARCH LABORATORIES
BENJAMIN FRANKLIN PARKWAY • PHILADELPHIA, PENNA. 19103

Technical

Final Report
F-B2294

Report

ON THE EIGENVALUES OF COUETTE
FLOW IN A FULLY-FILLED
CYLINDRICAL CONTAINER

by

M. M. Reddi

January 1967

Prepared for

EXTERIOR BALLISTIC LABORATORY
Aberdeen Proving Ground, Maryland

Contract No. DA-30-069-AMC-686(R)



THE FRANKLIN INSTITUTE RESEARCH LABORATORIES
BENJAMIN FRANKLIN PARKWAY • PHILADELPHIA, PENNA. 19103

SUMMARY

For a stationary flow in a cylindrical container of the Couette type in an outer radial zone, and of zero velocity in an inner radial zone, the normal mode equations are derived. For negative wave numbers in the θ -direction, these equations are found to have a singularity.

The eigenvalues are calculated by initial value methods employing the Runge-Kutta-Gill integration procedure. Values of the dependent function and their derivatives at the singularity are calculated by linear extrapolation coupled with continuity requirements.

Tables of eigenvalues for various slenderness ratios of the cylinder and various radial nodes are given for θ -wave numbers of -1 .

M. M. Reddi

M. M. Reddi
Senior Staff Engineer

Approved by:

Z. Zudans

Z. Zudans, Technical Director
Mechanical and Nuclear
Engineering Department



TABLE OF CONTENTS

	<u>Page</u>
SUMMARY.	i
INTRODUCTION	1
FORMULATION OF THE EIGENVALUE PROBLEM.	1
NORMAL MODES	6
Axisymmetric Cases.	9
Case i, The Narrow Gap Approximation for $m = 0$	10
Case ii, The Formal Solution for $m = 0$	13
THE GENERAL CASE	14
NUMERICAL SOLUTION	15
DESCRIPTION OF THE COMPUTER PROGRAM.	16
RESULTS.	20
APPENDIX	A

LIST OF FIGURES

<u>Fig. No.</u>		<u>Page</u>
1	Integration Intervals.	17

LIST OF TABLES

<u>Table No.</u>		<u>Page</u>
1	Input Map.	19
2	Effect of Step Size on Computed Eigenvalues.	20
3	Lowest Eigenvalues	21

INTRODUCTION

The object of the following investigation is to determine the eigenfrequencies of a viscous fluid, contained in a rotating cylinder, during the initial spin-up period. In order to make the problem tractable, an inviscid fluid is assumed and the fluid is divided into two zones; an outer radial zone in which a Couette type flow prevails and an interior zone which is static. The initial flow distribution, where the interface between the two zones is assumed to decrease from the outer radius to zero in a quasi-static manner, is assumed to be time independent.

FORMULATION OF THE EIGENVALUE PROBLEM

In the following we are interested in the characteristic frequencies of an inviscid liquid contained in a cylindrical container. The liquid is assumed to have an initial stable motion of the Couette type for an outer radial zone whereas the interior is assumed to be at zero velocity. Thus, let the container be of radius a and height $2c$. Then the stationary flow in the outer zone is given by

$$\begin{aligned} u_o &= 0 & a \geq r \geq b \\ v_o &= a\omega[(r/a) - (e^2 a/r)]/(1 - e^2) & 0 \leq [e = b/a] \leq 1 \\ w_o &= 0 \end{aligned} \quad (1)$$

In the inner zone,

$$\begin{aligned} u_i &= 0 \\ v_i &= 0 & b \geq r \geq 0 \\ w_i &= 0 \end{aligned} \quad (2)$$

where b is the radius of the interface between the static interior and the moving exterior; u , v , and w are the radial, longitudinal and axial components of velocity in a cylindrical coordinate system with the

z-axis aligned along the axis of the cylinder. The origin of the coordinate system is located at the bottom of the cylinder so that the z-coordinates of the end faces are given by 0 and $2c$. The velocity components in the inner and outer zones are distinguished by the subscripts i and o respectively.

For convenience, we consider the two zones, namely, the exterior zone of Couette flow, and the interior zone of zero velocity, separately. These are then coupled by boundary conditions imposed at the interface, where the radial velocities and pressures corresponding to the two regions are required to be equal. Considering the exterior zone first, the continuity and momentum equations with the appropriate boundary conditions are:

$$\begin{aligned}
 \frac{\partial u_o}{\partial t} + (Q_o \cdot \nabla) u_o - v_o^2/r &= - \frac{\partial}{\partial r} \left(\frac{p_o}{\rho} \right) \\
 \frac{\partial v_o}{\partial t} + (Q_o \cdot \nabla) v_o + u_o v_o/r &= - \frac{1}{r} \frac{\partial}{\partial \theta} \left(\frac{p_o}{\rho} \right) \\
 \frac{\partial w_o}{\partial t} + (Q_o \cdot \nabla) w_o &= - \frac{\partial}{\partial z} \left(\frac{p_o}{\rho} \right) \\
 Q_o \cdot \nabla &\equiv u_o \frac{\partial}{\partial r} + \frac{v_o}{r} \frac{\partial}{\partial \theta} + w_o \frac{\partial}{\partial z} \\
 \frac{u_o}{r} + \frac{\partial u_o}{\partial r} + \frac{1}{r} \frac{\partial v_o}{\partial \theta} + \frac{\partial w_o}{\partial t} &= 0
 \end{aligned} \tag{3}$$

The boundary conditions are

$$\begin{aligned}
 r &= a; u_o = 0 \\
 z &= 0, 2c; w_o = 0
 \end{aligned}$$

Moreover, at the interface, we have

$$u_{of} = u_{if}; p_{of} = p_{if} \tag{4}$$

where subscript f refers to the interface.

Substituting the following,

$$\begin{aligned} u_o^* &= u_o/a\omega; \quad v_o^* = v_o/a\omega; \quad w_o^* = w_o/a\omega \\ r^* &= r/at; \quad z^* = z/at; \quad t^* = \omega t; \quad \theta^* = \theta \\ p^*/\rho^* &= p/\rho a^2 \omega^2 \end{aligned} \quad (5)$$

and dropping the asterisk, the stationary flow is given by

$$\begin{aligned} u_o &= 0 \\ v_o &= [r - e^2/r]/[1 - e^2] \quad 1 \geq r \geq e \\ w_o &= 0 \end{aligned}$$

The boundary conditions now become

$$\begin{aligned} r &= 1; \quad u_o = 0 \\ z &= 0, \quad 2c/a; \quad w_o = 0 \end{aligned} \quad (6)$$

At the interface, they are

$$u_{of} = u_{if}; \quad p_{of} = p_{if} \quad (7)$$

Here u_{if} and p_{if} are the appropriate non-dimensional radial velocity and pressure prevailing in the flow field of the inner zone at the interface.

From the non-dimensional form of Equation 3,

$$\begin{aligned} \frac{\partial P_o}{\partial r} &= \frac{v^2}{r} \\ \frac{\partial P_o}{\partial \theta} &= 0 \\ \frac{\partial P_o}{\partial z} &= 0 \end{aligned} \quad (8)$$

where

$$\begin{aligned} P_o &= p_o/\rho \\ v &= [r - e^2/r]/(1 - e^2) \end{aligned} \quad (9)$$

Integrating,

$$P_o = \int_e^r (v^2/y) dy + P_1 \quad (10)$$

where P_1 is the equilibrium pressure at the interface.

We now assume perturbations due to a radial displacement, η , of the interface from its equilibrium position corresponding to $r = e$.

$$\begin{aligned} u_o &= u'_o \\ v_o &= V + v'_o \\ w_o &= w'_o \\ P_o &= \int_e^r \frac{V^2(y)}{y} dy + P'_o + P_1 \end{aligned} \quad (11)$$

Substituting the perturbations into the dimensionless Euler's equations of motion, and neglecting second and higher order quantities,

$$\begin{aligned} \frac{\partial u'_o}{\partial t} + \frac{V}{r} \frac{\partial u'_o}{\partial \theta} - \frac{2V}{r} v'_o &= - \frac{\partial P'_o}{\partial r} \\ \frac{\partial v'_o}{\partial t} + u'_o \frac{dV}{dr} + \frac{V}{r} \frac{\partial v'_o}{\partial \theta} + \frac{u'_o V}{r} &= - \frac{1}{r} \frac{\partial P'_o}{\partial \theta} \\ \frac{\partial w'_o}{\partial t} + \frac{V}{r} \frac{\partial w'_o}{\partial \theta} &= - \frac{\partial P'_o}{\partial z} \\ \frac{u'_o}{r} + \frac{\partial u'_o}{\partial r} + \frac{1}{r} \frac{\partial v'_o}{\partial \theta} + \frac{\partial w'_o}{\partial z} &= 0 \end{aligned} \quad (12)$$

The boundary conditions become

$$\begin{aligned} r = 1; u'_o &= 0 \\ z = 0, 2c/a; w'_o &= 0 \end{aligned} \quad (13)$$

The boundary conditions due to displacement of the interface are obtained in the following manner:

Let the interface be at

$$r = e + \eta \quad (14)$$

Substituting into the last of Equations 11 and neglecting second and higher order terms,

$$P_{of} = P_1 + P'_{of} \quad (15)$$

At the interface, we have from 14

$$u'_0 = \frac{\partial \eta}{\partial t} \quad (16)$$

where second and higher order quantities in the perturbations have been ignored.

We now consider the interior zone. The Euler's equations of motion and the equation of continuity are once again given by Equation 3 in a non-dimensional form. The boundary conditions are

$$z = 0, 2c/a; w_1 = 0 \quad (17)$$

$$r = 0; u_1, v_1 \text{ and } w_1 \text{ are bounded.}$$

At the interface, they are

$$\begin{aligned} u_{1f} &= u_{of} \\ P_{1f} &= P_{of} \end{aligned} \quad (18)$$

We once again assume perturbations due to radial displacement, η , of the interface from its equilibrium position corresponding to $r = e$.

$$\begin{aligned} u_1 &= u'_1 \\ v_1 &= v'_1 \\ w_1 &= w'_1 \\ P_1 &= P'_1 + P_1 \end{aligned} \quad (19)$$

Substituting in Equation 3,

$$\begin{aligned} \frac{\partial u'_1}{\partial t} &= - \frac{\partial P'_1}{\partial r} \\ \frac{\partial v'_1}{\partial t} &= - \frac{1}{r} \frac{\partial P'_1}{\partial \theta} \\ \frac{\partial w'_1}{\partial t} &= - \frac{\partial P'_1}{\partial z} \\ \frac{u'_1}{r} + \frac{\partial u'_1}{\partial r} + \frac{1}{r} \frac{\partial v'_1}{\partial \theta} + \frac{\partial w'_1}{\partial z} &= 0 \end{aligned} \quad (20)$$

The boundary conditions become

$$z = 0, 2c/a, w'_1 = 0 \quad (21)$$

$$r = 0; u'_1, v'_1 \text{ and } w'_1 \text{ are bounded.}$$

At the interface

$$\begin{aligned} u'_1 &= \frac{\partial \eta}{\partial t} \\ P'_{if} &= P_{of} - P_1 \end{aligned} \quad (22)$$

NORMAL MODES

In accordance with the usual procedure of treating characteristic value problems, the perturbations are analyzed into normal modes. In view of the boundary conditions, it is natural to suppose that the perturbations are given by quantities which have a (r, θ, z, t) dependence given by

$$\begin{aligned} u'_0 &= U_0(r) \cos [h_0 \pi a z / 2c] e^{i(K_0 t + m_0 \theta)} \\ v'_0 &= B_0(r) \cos [h_0 \pi a z / 2c] e^{i(K_0 t + m_0 \theta)} \\ w'_0 &= W_0(r) \sin [h_0 \pi a z / 2c] e^{i(K_0 t + m_0 \theta)} \\ P'_0 &= R_0(r) \cos [h_0 \pi a z / 2c] e^{i(K_0 t + m_0 \theta)} \end{aligned} \quad (23)$$

Where K_0 is a constant (which can be complex), m_0 is an integer (which can be positive, zero, or negative) and h_0 is the wave number in the z -direction (which can be 1, 2, 3, etc.). U_0 , B_0 , W_0 and R_0 are functions of r only.

In an analogous manner, for the interior zone, the perturbation quantities may be assumed to be of the form

$$\begin{aligned}
u'_1 &= U_1(r) \cos(h_1 \pi a z / 2c) e^{i(K_1 t + m_1 \theta)} \\
v'_1 &= B_1(r) \cos(h_1 \pi a z / 2c) e^{i(K_1 t + m_1 \theta)} \\
w'_1 &= W_1(r) \sin(h_1 \pi a z / 2c) e^{i(K_1 t + m_1 \theta)} \\
p'_1 &= R_1(r) \cos(h_1 \pi a z / 2c) e^{i(K_1 t + m_1 \theta)}
\end{aligned} \tag{24}$$

where K_1 , m_1 , and h_1 are defined as before. U_1 , B_1 , W_1 , and R_1 are once again assumed to be functions of the radius r only.

Substituting Equation 23 into Equation 12, we obtain

$$\begin{aligned}
i(K_o + m_o \frac{V}{r})U_o - \frac{2V}{r} B_o &= -\frac{dR_o}{dr} \\
(\frac{dV}{dr} + \frac{V}{r})U_o + i(K_o + m_o \frac{V}{r})B_o &= -i m_o \frac{R_o}{r} \\
(K_o + m_o \frac{V}{r})W_o &= -\frac{h_o \pi a}{2c} i R_o \\
(\frac{dU_o}{dr} + \frac{U_o}{r}) + i m_o \frac{B_o}{r} + \frac{h_o \pi a}{2c} W_o &= 0
\end{aligned} \tag{25}$$

In a similar manner, substituting Equation 24 into 20, we obtain

$$\begin{aligned}
U_1 &= \frac{i}{K_1} \frac{dR_1}{dr} \\
B_1 &= -\frac{1}{r} \frac{m_1}{K_1} R_1 \\
W_1 &= -\frac{i}{K_1} \frac{h_1 \pi a}{2c} R_1 \\
\frac{U_1}{r} + \frac{dU_1}{dr} + \frac{i m_1}{r} B_1 + \frac{h_1 \pi a}{2c} W_1 &= 0
\end{aligned} \tag{26}$$

Since

$$u'_{of} = u'_{if}$$

we have

$$\begin{aligned}
 U_1(e + \eta) \cos(h_1 \pi a z / 2c) e^{i(K_1 t + m_1 \theta)} \\
 = U_0(e + \eta) \cos(h_0 \pi a z / 2c) e^{i(K_0 t + M_0)}
 \end{aligned}
 \quad (27)$$

Inasmuch as θ , z and t are independent variables, we require that K_1 , m_1 and h_1 be equal to K_0 , m_0 and h_0 respectively. Thus, the subscripts for K , m and h will be omitted in what follows. Equation 25 can be simplified to the following

$$\begin{aligned}
 \sigma \left[\frac{dU_0}{dr} + \frac{U_0}{r} \right] - \frac{m}{r^2} U_0 \frac{d}{dr} (\Omega r^2) &= \left[\frac{m^2}{r^2} + \frac{\pi^2 h^2 a^2}{4c^2} \right] i R_0 \\
 \sigma^2 U_0 - \frac{2\Omega}{r} U_0 \frac{d}{dr} (\Omega r^2) &= \left[2\Omega m \frac{R_0}{r} + \sigma \frac{dR_0}{dr} \right] i
 \end{aligned}
 \quad (28)$$

where

$$\Omega = V/r \quad (29)$$

and

$$\sigma = K + m\Omega \quad (30)$$

In a similar manner, Equation 26 becomes

$$\begin{aligned}
 \frac{U_1}{r} + \frac{dU_1}{dr} &= \frac{1}{K} \left[\frac{m^2}{r^2} + \frac{\pi^2 h^2 a^2}{4c^2} \right] R_1 \\
 U_1 &= \frac{1}{K} \frac{dR_1}{dr}
 \end{aligned}
 \quad (31)$$

The boundary conditions are

$$U_0(1) = 0 \quad (32)$$

$$U_1(0) \text{ and } R_1(0) \text{ are bounded}$$

The interface conditions are obtained as follows: Since P_{of} is equal to P_{if} , from Equations 15 and 22,

$$P'_{if} = P'_{of} \quad (33)$$

Substituting for P'_{of} and P'_{if} from Equations 23 and 24 respectively,

$$\begin{aligned} R_1(e) \cos(h\pi az/2c) e^{i(Kt + m\theta)} \\ = R_0(e) \cos(h\pi az/2c) e^{i(Kt + m\theta)} \end{aligned} \quad (34)$$

where we consider $\eta \ll e$.

$$\left[K \left\{ \frac{U_1}{r} + \frac{dU_1}{dr} \right\} = \sigma \left\{ \frac{dU_0}{dr} + \frac{U_0}{r} \right\} - \frac{m}{r^2} U_0 \frac{d}{dr} (\Omega r^2) \right]_{r=e} \quad (35)$$

$$U_0(e) = U_1(e) \quad (36)$$

Thus, the characteristic value problem consists of Equations 28 and 31 with boundary conditions 32, 35, and 36. Since we are interested only in those motions which produce a couple on the casing, interest centers around values of $h = (2j + 1)$, where j is an integer.

Axisymmetric Cases^{*}

Setting $m = 0$ reduces Equations 28 and 31 to the following

$$\begin{aligned} \frac{d^2 U_0}{dr^2} + \frac{1}{r} \frac{dU_0}{dr} - \left[\frac{1}{r^2} + N^2 - \frac{N^2}{K^2} \frac{2\Omega}{r} \frac{d}{dr} (\Omega r^2) \right] U_0 &= 0 \\ \frac{d^2 U_1}{dr^2} + \frac{1}{r} \frac{dU_1}{dr} - \left[\frac{1}{r^2} + N^2 \right] U_1 &= 0 \end{aligned} \quad (37)$$

$$\text{where } N^2 \equiv \pi^2 h^2 a^2 / 4c^2 \quad (38)$$

The boundary conditions become

$$\begin{aligned} U_0(1) &= 0 \\ U_0(e) &= U_1(e); \quad \frac{dU_0(e)}{dr} = \frac{dU_1(e)}{dr} \\ U_1(0) &= 0 \end{aligned} \quad (39)$$

^{*}The axisymmetric case is of academic interest only.

Case 1, The Narrow Gap Approximation for $m = 0$

In specifying Ω , interest is in those velocity distributions which are realizable in a viscous fluid. Equations 9 and 29 determine the case of primary interest as

$$\Omega = (1 - e^2/r^2)/(1 - e^2) \quad (40)$$

However, as $e \rightarrow 1$, an important simplification in the characteristic value problem given by Equations 37, 39 and 40 is possible provided that $(1 - e)$ is small when compared with $(1 + e)/2$. In this case, Equation 40 may be expressed as

$$\Omega = s \quad (41)$$

$$s = (r - e)/(1 - e) \quad (42)$$

The first of Equation 37 may be expressed as

$$\left[\frac{d^2}{dr^2} + \frac{1}{r} \frac{d}{dr} - \frac{1}{r^2} - N^2 \right] U_o = - \frac{N^2}{K^2} \left[2\Omega r \frac{d\Omega}{dr} + 4\Omega^2 \right] U_o \quad (43)$$

Using the transformation defined by Equation 42, consistent with the "narrow gap" approximation of small $(1 - e)$, equation 43 becomes

$$\left[\frac{d^2}{ds^2} - a_1^2 \right] U_o = - \frac{a_1^2}{K^2} \cdot \frac{2e}{1-e} s U_o \quad (44)$$

where

$$a_1^2 \equiv (1 - e)^2 N^2 \quad (45)$$

By means of the following transformation,

$$x = a_1^{2/3} b^{2/3} \left[\frac{1}{b^2} - s \right] \quad (46)$$

where

$$b^2 \equiv 2e/K^2(1 - e) \quad (47)$$

Equation 44 may be expressed as

$$\frac{d^2 U_o}{dx^2} - x U_o = 0 \quad (48)$$

The general solution of Equation 48 may be given in terms of either Bessel Functions of order $1/3$ or somewhat more conveniently in terms of Airy's functions $AI(x)$ and $BI(x)$; thus,

$$U_o = A_o AI(x) + B_o BI(x) \quad (49)$$

where A_o and B_o are the integration constants.

In order to determine U_i , the general solution of the second of the differential Equation 37 may be obtained from

$$\frac{d^2 U_i}{dR^2} + \frac{1}{R} \frac{dU_i}{dR} - \left[\frac{1}{R^2} + 1 \right] U_i = 0 \quad (50)$$

$$\text{where } R^2 = N^2 r^2 \quad (51)$$

The general solution of Equation 50 may be written as

$$U_i = A_i I_1(R) + B_i K_1(R) \quad (52)$$

The boundary conditions in terms of the new independent variable x defined by Equation 46 are

$$U_o(x_1) = 0 \quad (53a)$$

$$U_o(x_2) = U_i(Ne) \quad (53b)$$

$$-\frac{dU_o}{dx} = \frac{N(1-e)}{a_1^{2/3} b^{2/3}} \frac{dU_i}{dR} \text{ at } x = x_2; R = Ne \quad (53c)$$

$$U_i(0) = 0 \quad (53d)$$

where

$$\begin{aligned} x_1 &\equiv a_1^{2/3} b^{2/3} \left[\frac{1}{b^2} - s \right] \\ x_2 &\equiv a_1^{2/3} b^{2/3} / b^2 \end{aligned} \quad (54)$$

From boundary conditions 53d, we have

$$A_1 I_1(o) + B_1 K_1(o) = 0$$

Therefore,

$$B_1 = 0 \quad (55)$$

Substituting the general solutions given by Equations 49 and 52 and the condition given by Equation 55 into the boundary conditions 53a, 53b and 53c, we obtain

$$A_o AI(x_1) + B_o BI(x_1) = 0 \quad (56)$$

$$A_o AI(x_2) + B_o BI(x_2) = A_1 I_1(Ne) \quad (57)$$

$$\begin{aligned} & - \frac{a_1^{2/3} b^{2/3}}{a_1} \left[A_o AI'(x_2) + B_o BI'(x_2) \right] \\ & = A_1 \left[I_o(Ne) - \frac{I_1(Ne)}{Ne} \right] \end{aligned} \quad (58)$$

where the primes refer to differentiation with respect to x . A_o , B_o and A_1 may be eliminated from Equations 56, 57 and 58 to give the characteristic equation as

$$x_2^{-1/2} \left[\frac{AI'(x_2) BI(x_1) - BI'(x_2) AI(x_1)}{AI(x_1) BI(x_2) - AI(x_2) BI(x_1)} \right] = \frac{I_o(Ne)}{I_1(Ne)} - \frac{1}{Ne} \quad (59)$$

From the definition of x_1 , and x_2 in Equation 54, we obtain

$$x_1 = x_2 - N(1 - e)/x_2^{1/2} \quad (60)$$

For given N and e , the values of x_1 and x_2 which satisfy Equations 59 and 60 simultaneously are the desired values. However, not all pairs of x_1 and x_2 lead to admissible values of K^2 . From Equation 61 we observe that since

$$K^2 = \frac{2e}{1-e} \cdot \frac{x_2}{x_2 - x_1} = \frac{2e}{N(1-e)^2} x_2^{3/2} \quad (61)$$

and since K^2 is a real number, first of all x_2 must be a positive number. Secondly, $(x_2 - x_1)$ must also be positive. Thus, of all the values which satisfy Equations 59 and 60, we select only those pairs of x_1 and x_2 values which satisfy the conditions that $x_2 > 0$ and that $x_2 > x_1$.

Case ii, The Formal Solution for $m = 0$

In case we wish to consider the complete range of e in $1 \geq e \geq 0$, we must consider Equation 37 without making any approximation. We then have

$$\frac{d^2 U_o}{dr^2} + \frac{1}{r} \frac{dU_o}{dr} - \left[\frac{v^2}{r^2} + \alpha^2 \right] U_o = 0 \quad (62)$$

where

$$v^2 \equiv 1 + \frac{4N^2 e^2}{K^2 (1 - e^2)^2} \quad (63)$$

$$\alpha^2 \equiv N^2 - \frac{4N^2}{K^2 (1 - e^2)^2} \quad (64)$$

Equation 62 would need to be considered with the second of Equation 37 and boundary conditions 39. The general solution of Equation 62 is

$$U_o = \ell_v(\alpha r) \quad (65)$$

where ℓ is a general cylinder function of order v . The boundary conditions require ℓ_v to vanish at $r = 1$. The required solution may be expressed as

$$U_o = M [J_{-\nu}(\alpha) J_{\nu}(\alpha r) - J_{\nu}(\alpha) J_{-\nu}(\alpha r)] \quad (66)$$

where M is a constant.

Defined in this manner, U_o clearly vanishes at $r = 1$. We have further at $r = e$.

$$M[J_{-\nu}(\alpha) J_{\nu}(\alpha e) - J_{\nu}(\alpha) J_{-\nu}(\alpha e)] = A_1 I_1(Ne) \quad (67)$$

Moreover,

$$\left. \frac{dU_o}{dr} \right|_{r=e} = A_1 N \left[I_o(Ne) - \frac{I_1(Ne)}{Ne} \right] \quad (68)$$

The integration constants M and A_1 ; may be eliminated from Equations 67 and 68 to yield the characteristic equation. For arbitrarily assigned ν , one may compute K^2 from the characteristic equation and Equations 63 and 64 for given N and e.

THE GENERAL CASE

Consistent with boundedness of the solutions of U_i and R_i at the origin Equation 31 has the following solution

$$R_i = A I_m(Nr) \quad (69)$$

$$U_i = \frac{1}{K} R'_i = A \left[\frac{m}{r} I_m + N I_{m+1} \right] \quad (70)$$

where A is any arbitrary constant. Equation 28 may be rewritten as

$$U'_o = \frac{1}{\sigma} \left[\left(\frac{m^2}{r^2} + N^2 \right) R_o + \frac{2m}{1-e^2} \cdot \frac{1}{r} U_o \right] - \frac{U_o}{r} \quad (71)$$

$$R'_o = -\frac{2}{\sigma} \left[\frac{2}{1-e^2} U_o + \frac{m}{r} R_o \right] + \sigma U_o \quad (72)$$

The boundary conditions become

$$U_o(e) = \frac{A}{K} \left[\frac{m}{e} I_m(Ne) + N I_{m+1}(Ne) \right] \quad (73)$$

$$R_o(e) = A I_m(Ne) \quad (74)$$

$$U_o(1) = 0 \quad (75)$$

where R_o is redefined as $-iR_o$ of equations prior to Equation 69. The eigenvalue problem now is the following: find those values of K for which the Equations 71 and 72, and the boundary conditions 73, 74, and 75 are satisfied.

Since Equations 71 and 72 are homogeneous, any constant times any solution that satisfies the boundary conditions will also be a solution. Thus, in the eigenvalue problem, we can arbitrarily assign any value to the constant A . For convenience, we set it equal to unity.

NUMERICAL SOLUTION

Since Equations 71 and 72 are too cumbersome for closed solutions, we employ numerical integration methods. The numerical procedure for solving the eigenvalue problem is as follows: for an assumed value of K , we can compute the values of $U_o(e)$ and $R_o(e)$ for a given set of values of m , e , and N . Using these as the initial values, we can integrate Equations 71 and 72 by numerical methods to yield $U_o(1)$. If this value is zero, then our choice of K is indeed the right one. If not, we change our assumed value of k by a selected increment and recompute. In practice, since it is uneconomical, computer time-wise, to calculate a precise zero value for $U_o(1)$, we arbitrarily say that if the absolute value of $U_o(1)$ is less than a prescribed positive small number, then our choice of the k value is the right one.

For negative values of m , Equations 71 and 72 have a singularity depending on whether or not the quantity $1 + (1 - e^2) K/m$ is real or

complex. If the quantity is real, the singularity is at a radius given by

$$r_{\text{singular}} = e / \sqrt{1 + (1 - e^2) K/m} \quad (76)$$

At this radius, the derivatives of U_0 and R_0 cannot be computed from Equations 71 and 72. However, from physical considerations, we require that U_0 and R_0 be continuous at this point.

When a singularity exists, we employ the following procedure: We divide the two zones, $e \leq r \leq r_{\text{sing}}$ and $r_{\text{sing}} \leq r \leq 1$ into N_1 and N_2 intervals, where N_1 and N_2 are selected to be such that the increments in radius for an interval is approximately equal in both zones. Starting from $r = e$, we integrate the Equations 71 and 72 till we reach the last interval of $e \leq r \leq r_{\text{sing}}$. We subdivide this into four smaller intervals and integrate the equations to obtain the values of U_0 , R_0 and their derivatives at the three intermediate points shown in Figure 1. Using these values, by linear extrapolation, we estimate the values of U_0 , R_0 and their derivatives at the singularity. Since we cannot use Equations 71 and 72 at the singularity, using the estimated derivatives, we integrate numerically to the one-fourth point of the next interval. From here on, we revert to the conventional integration procedure using Equations 71 and 72.

The numerical integration procedure used in the computer program for calculating the eigenvalues is the familiar Runge-Kutta-Gill method.

DESCRIPTION OF THE COMPUTER PROGRAM

For computing an eigenvalue, the input data to the program is supplied by two cards. The input map for the program, describing the breakdown of the entries in the cards are shown in Table 1. The third I-field on the first card of each set was originally intended to supply the value of n . Since, this was subsequently generalized to be a

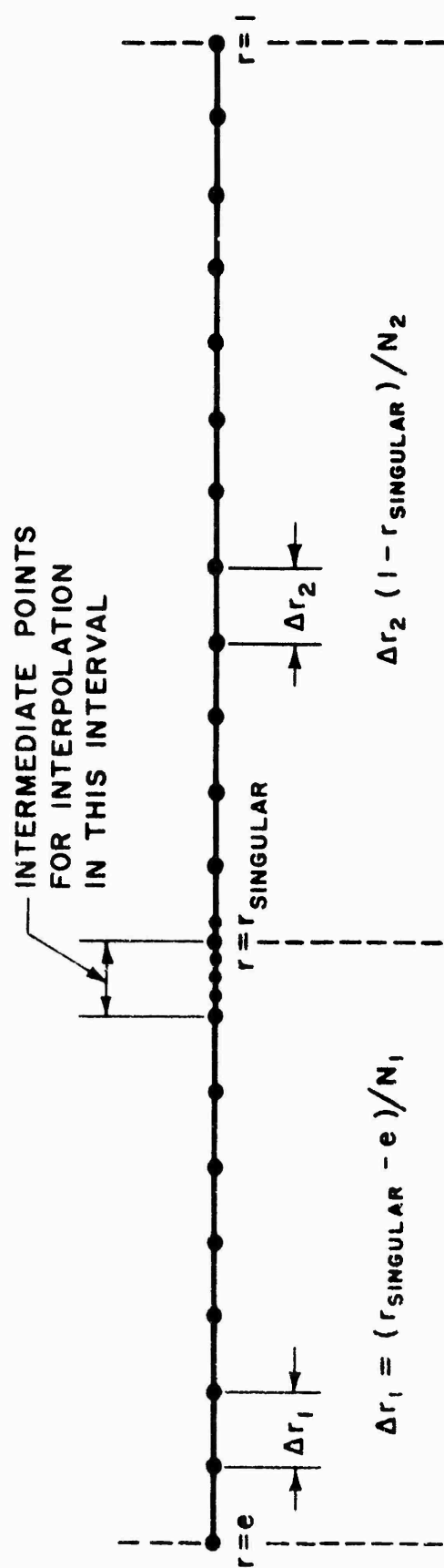


FIG. 1. INTEGRATION INTERVALS

Table 1
INPUT MAP

	<u>Variable</u>	<u>Definition</u>	<u>Format</u>
Card No. 1	M	Run Number	I6
	MM	-m	I6
	NN	Enter zeros	I6
	NR	Number of integration intervals ≤ 500	I6
	IGEN	Number of Eigenvalues required	I6
	NSTOP	If last set of data being read, should be entered non- zero. Otherwise, enter zero.	I6
	LC	10 Ne ≤ 50	I6
Card No. 2	PLAO	Initial trial value of Eigen- value	E12.0
	DPLA	Increment of Eigenvalue	E12.0
	TRUNC	Permissible truncation difference	E12.0
	CRAD	e	F12.6

factor in N (CAPN in the program), it may be filled in with a zero or any other number. This number will appear on the printout as the value of n (N in the program). Since it has no other effect on the program, it may be ignored for all other purposes.

The computer program, written in Fortran IV, is given in the Appendix: the first subroutine, NING(NR1) provides the Runge-Kutta-Gill integration method. It also calls a subroutine, DER(Y, FM, CØN, CRAD, FK, AK) which provides the derivatives of U_0 and R_0 as given by Equations 71 and 72. Subroutine NUESTI, called by the main program, provides a method for incrementing the eigenvalue on the basis of current derivation in $U_0(1)$ from the specified truncation error. Subroutine RKGÇØN (AIN, BIN, CIN) supplies the RKG constants to the main program. Subroutine BESCØN(AI, BI), called by the main program supplies the values of the modified Bessel functions. If the program is to be run for any values of m other than -1, this subroutine will need to be changed to supply the modified Bessel functions of the appropriate order.

RESULTS

Table 2 shows the effect of various step sizes on the computed eigenvalues. Truncation is of 10^{-6} , initial eigenvalue is 0.01, eigenvalue increment is .05.

Table 2
EFFECT OF STEP SIZE ON COMPUTED EIGENVALUES

Total No. of Steps	Lowest eigenvalue K for Ne = .2 b/a = e = 0.45 m = -1	Lowest eigenvalue K for Ne = .2 b/a = e = .1 m = -1
100	.76079984	.35125716
200	.75407226	.35046438
300	.75447285	.35032046
400	.75445469	.35025621
500	.75479767	.35021697

The results show that if 200 or more total number of steps are employed, the variation is in the fourth significant digit.

Table 3 lists the results for various Ne and e values computed with 200 integration steps, an initial eigenvalue of 0.01, and an eigenvalue increment of 0.05. With this sweep procedure, the listed eigenvalues were found to be the lowest. It is possible that if the eigenvalue increment is lowered from 0.05 to some smaller value, other, ever lower, eigenvalues may be discovered; but this is considered to be extremely unlikely.

In Stewartson's^{*} notation, we have

$$\frac{c}{a(2j+1)} = \frac{\pi}{2N}$$

$$b/a = e$$

$$\tau = K$$

^{*} Stewartson, K., *On The Stability of a Spinning Top Containing Liquid*, J. Fluid Mech., 5, 1959, pp. 577-592.

Table 3
LOWEST EIGENVALUES

10XNe	$\frac{\pi}{2} \frac{a(2j+1)}{c}$	b/a	K = τ	n	10XNe	$\frac{\pi}{2} \frac{a(2j+1)}{c}$	b/a	K = τ	n
1	2	.05	.35294417	2	3	6	.05	0.10161652	4
	1	.1	.36637878	1		3	.1	0.079672609	2
	.5	.2	.68914656	1		1.5	0.2	0.077654683	1
	.333	.3	.78440604	1		1	0.3	0.38693088	1
	.25	.4	.95493352	1		.75	0.4	0.53024257	1
	.2	.5	.98246679	1		.6	0.5	0.65401489	1
	.166	.6	.99240964	1		.5	0.6	0.93675748	1
	.142857	.7	.99677785	1		.42871	0.7	0.96985291	1
	.125	.8	.99793433	1		.375	0.8	0.98589012	1
	.1111	.9	.99749419	1		.333	0.9	0.99401352	1
2	4	.05	.17450405	3	4	8	.05	0.062517198	5
	2	.1	.35046438	2		4	.1	0.1748773	3
	1	.2	.37567224	1		2	.2	0.33798441	2
	.666	.3	.57999146	1		1.33	.3	0.19826175	1
	.5	.4	.68582012	1		1	0.4	0.39889403	1
	.4	.5	.93226799	1		0.8	0.5	0.5155985	1
	.333	.6	.97111366	1		0.6667	0.6	0.69132881	1
	.285714	.7	.98598000	1		0.571429	0.7	0.94881139	1
	.25	.8	.99387097	1		0.5	0.8	0.97572948	1
	.222	.9	.99698953	1		0.444	0.9	0.98943233	1

n is the number of radial nodes in the range $0 < r < 1$

Table 3 (Concl.)

$10XNe$	$\frac{\pi}{2} \frac{a(2i+1)}{c}$	b/a	$k = \tau$	n	$10XNe$	$\frac{\pi}{2} \frac{a(2i+1)}{c}$	b/a	$k = \tau$	n
5	10	.05	0.03821307	6	7	14	.05	.094461123	9
	5	.1	0.021487953	3		7	.1	.15244942	5
	2.5	0.2	0.21049394	2		3.5	.2	.24648402	3
	1.666	0.3	0.028242248	1		2.333	.3	.24666304	2
	1.25	0.4	0.27032753	1		1.75	.4	.036690115	1
	1	0.5	0.41369377	1		1.4	.5	.23270392	1
	.833	0.6	0.54600539	1		1.166	.6	.36577586	1
	.714268	0.7	0.9210945	1		1	.7	.53633700	1
	.625	0.8	0.96316761	1		.875	.8	.93412168	1
	.555	0.9	0.98512214	1		.777	.9	.97391060	1
6	12	.05	0.021654819	7	8	16	.05	.076405596	10
	6	.1	0.1034529	4		8	.1	.065089666	5
	3	.2	0.094563088	2		4	.2	.17042725	3
	2	.3	0.32489160	2		2.666	.3	.17456094	2
	1.5	.4	0.14853735	1		2	.4	.32350274	2
	1.2	.5	0.32126868	1		1.6	.5	.14869081	1
	1	.6	0.44177624	1		1.333	.6	.29887522	1
	.857143	.7	0.69269504	1		1.142857	.7	.44032151	1
	.75	.8	0.94963371	1		1	.8	.91714270	1
	.666	.9	0.97912326	1		.888	.9	.96724922	1

n is the number of radial nodes in the range $0 < r < 1$

APPENDIX

A

MAIN PROGRAM FOR THE DETERMINATION OF EIGENVALUES 31G-B2294-01

```

COMMON FM,FN,CON,DEL,NR,X(1,501),Z(1,501),FREQU(3),DET(3),AIN(4),
1BIN(4),CIN(4),CRAD,DPLA,IDET,ILU,KLUE,UCO,RCO,RADIUS(501)
DIMENSION AI(50),BI(50)
NSTOP=0
CALL RKGCON(AIN,BIN,CIN)
CALL BESCON(AI,BI)
20 IF(NSTOP.NE.0)STOP
READ(5,10)M,MM,NN,NR,IGEN,NSTOP,LC,PLAO,DPLA,TRUNC,CRAD
10 FORMAT(7I6/3E12.0,F12.6)
MM=-MM
FM=MM
FN=NN
FNR=NR
FREQU(1)=PLAO
BNC=LC
FNC=BNC/10.
CAPN=FNC/CRAD
CON=CAPN**2
AIM1=AI(LC)*EXP(FNC)
AIM2=BI(LC)*EXP(FNC)
IDET=1
KKK=1
WRITE(6,85)
85 FORMAT(1H1 35HCURRENT VALUES OF INTEGRATION DATA //)
WRITE(6,86)CAPN,LC
86 FORMAT(5X7HCAPN = F10.4,5X5HLC = I2)
DO 30 IG=1,IGEN
ICOUNT=1
70 UCO=(FM*AIM1/CRAD+CAPN*AIM2)/FREQU(1)
RCO=AIM1
RADIUS(1)=CRAD
X(1,1)=UCO
Z(1,1)=RCO
CALL NING(NR1)
DET(1)=X(1,NR+1)
WRITE(6,90)M,IG,FREQU(1),X(1,NR+1)
90 FORMAT(
// 10X 13HRUN NUM
1BER = I3, 5X 5HIG = I2, 5X 16HTRIAL EIGENV. = E15.8, 5X 9H U(1) 00000023
2= E15.8) 00000024
IF(IDET=2)170,170,180 00000025
170 DL = DET(1) 00000026
AL = FREQU(1) 00000027
GO TO 190 00000028
180 IF(AL=FREQU(1))170,170,190 00000029
190 ADET = ABS (DET(1)) 00000030
IF(ADET=TRUNC)80,80,100 00000031
100 CALL NUESTI 00000032
GO TO (80,110, 20),KLUE 00000035
110 IDET = IDET+1 00000036
ICOUNT=ICOUNT+1 00000037
IF(ICOUNT=50)70,70, 20
80 WRITE(6,1)
1 FORMAT(1H1 41X 32I EIGENVALUES AND EIGENFUNCTIONS ) 00000040

```

REDUI

04/01/66

MAIN

- EFN SOURCE STATEMENT - IFN(S) -

```

WRITE(6,2)M,MM,NN,IG,FREQU(1),CAPN,CRAD
2  FORMAT(//48X 10HR IN NO. = I3// 7X 6H M = I3, 5X 8H N = I3,
1  5X I2, 2X 13H EIGENVALUE = E15.8,9X 6H CAPN = F12.6,5X 4H C = F6.2)
WRITE(6,3)
3  FORMAT(//18X 2H R 4X 14H U MODE SHAPE 4X 14H R MODE SHAPE 18X 2H
5R 4X 14H U MODE SHAPE 3X 15H R MODE SHAPE //)
KPA=1
NRR=NR/2
DO 40 I=1,NRR
IF (I=51*KPA) 50,60,50
60 KPA = KPA+1
WRITE(6,1)
WRITE(6,2)M,MM,NN,IG,FREQU(1),CAPN,CRAD
WRITE(6,3)
50 K=NRR+1
WRITE(6,130) RADIUS(I),X(1,I),Z(1,I),RADIUS(K),X(1,K),Z(1,K)
130 FORMAT(F20.4, 3X E15.8, 3X E15.8,F20.4, 3X E15.8, 3X E15.8)
IF(I-NRR) 40,140,40
140 K = K+1
WRITE(6,150) RADIUS(K),X(1,K),Z(1,K)
150 FORMAT(40X F20.4,3X E15.8, 3X E15.8)
40 CONTINUE
FREQU(2)=AL
DFT(2) =DL
FREQU(1)=AL+DPLA
30 IDET=2
GO TO 20
END

```

00000047

00000048

00000049

00000051

00000055

00000057

00000058

00000059

00000062

00000063

00000064

00000065

00000066

00000067

00000068

```

SUBROUTINE NING(NR1)
COMMON FM,FN,CON,DEL,NR,X(1,501),Z(1,501),FREQU(3),DET(3),AIN(4),
1BIN(4),CIN(4),CRAD,DPLA,DET,ILU,KLUE,UCO,RCO,RADIUS(501)
DIMENSION AK(3),Y(3),Q(3),AM(3),BM(3)
FK=FREQU(1)
Y(1)=CRAD
Y(2)=UCO
Y(3)=RCO
Q(1)=0,
Q(2)=0,
Q(3)=0,
AK(1)=1,
BING=1.+(1.-(CRAD**2))*FK/FM
IF (BING)3,3,1
1 RSING=CRAD/SQRT(BING)
WRITE(6,2) RSING
2 FORMAT(/20X 14HR(SINGULAR) = F9.5)
GO TO 4
3 WRITE(6,5)
5 FORMAT(10X38HSINGULARITY OUTSIDE INTEGRATION DOMAIN )
4 BNR=NR
NR1=BNR*(RSING-CRAD)/(1.-CRAD)
NRM=NR-50
IF(NR1.LT.50)NR1=50
IF(NR1.GT.NRM)NR1=NRM
NR2=NR-NR1
FNR1=NR1
FNR2=NR2
DEL=(RSING-CRAD)/FNR1
DEL1=(1.-RSING)/FNR2
DELTA=DEL/4.
NAB=NR1+2
DO 1000 L=1,NAB
IF(L.GE.NR1)DEL=DELTA
DO 100 JJ=1,4
CALL DER(Y,FM,CON,CRAD,FK,AK)
DO 50 I=1,3
AIKN = AIN(JJ)*(AK(I)-BIN(JJ)*Q(I))
Y(I) = Y(I)*DEL*AIKN
50 Q(I) = Q(I)*3.*AIKN -CIN(JJ)*AK(I)
100 CONTINUE
RADIUS(L+1)=Y(1)
X(1,L+1)=Y(2)
1000 Z(1,L+1)=Y(3)
CALL DER(Y,FM,CON,CRAD,FK,AK)
BU=AK(2)
BR=AK(3)
Y(1)=RADIUS(NAB)
CALL DER(Y,FM,CON,CRAD,FK,AK)
AU=AK(2)
AR=AK(3)
AK(2)=2.*BU-AU
AK(3)=2.*BR-AR
Y(1)=RSING
Y(2)=2.*X(1,NAB+1)-X(1,NAB)

```

GOODY

- EFN SOURCE STATEMENT - IFN(S) -

```
Y(3)=2.*Z(1,NAB+1)-Z(1,NAB)
RADIUS(NR1+1)=Y(1)
X(1,NR1+1)=Y(2)
Z(1,NR1+1)=Y(3)
DEL=DEL1/4.
DO 80 L=1,4
DO 80 JJ=1,4
IF(L,EQ,1)GO TO 81
CALL DER(Y,FM,CON,GRAD,FK,AK)
81 DO 82 I=1,3
   AIKN =AIN(JJ)*(AK(I)-BIN(JJ)*Q(I))
   Y(I) = Y(I)+DEL*AIKN
82 Q(I) = Q(I)+3.*AIKN -CIN(JJ)*AK(I)
80 CONTINUE
RADIUS(NR1+2)=Y(1)
X(1,NR1+2)=Y(2)
Z(1,NR1+2)=Y(3)
DEL=DEL1
DO 900 L=NAB,NR
IF(L,EQ,NR)DEL=1,-Y(1)
DO 90 JJ=1,4
CALL DER(Y,FM,CON,GRAD,FK,AK)
DO 60 I=1,3
   AIKN =AIN(JJ)*(AK(I)-BIN(JJ)*Q(I))
   Y(I) = Y(I)+DEL*AIKN
60 Q(I) = Q(I)+3.*AIKN -CIN(JJ)*AK(I)
90 CONTINUE
RADIUS(L+1)=Y(1)
X(1,L+1)=Y(2)
900 Z(1,L+1)=Y(3)
RETURN
END
```


DERIVE - EFN SOURCE STATEMENT - IFN(S) -

```
SUBROUTINE DER(Y,FM,CON,CRAD,FK,AK)
DIMENSION Y(3),AK(3)
R2=1./(Y(1)*Y(1))
ALPHA=FM*FM*R2+CON
BETA=2./(1.-CRAD*CRAD)
OMEGA=.5*(1.-CRAD*CRAD*R2)*BETA
SIGMA=FK+FM*OMEGA
AK(2)=(ALPHA*Y(3)+FM*BETA*Y(2)/Y(1))/SIGMA-Y(2)/Y(1)
AK(3)=-2.*OMEGA*(BETA*Y(2)+FM*Y(3)/Y(1))/SIGMA+SIGMA*Y(2)
RETURN
END
```

```

SUBROUTINE NUESTI
COMMON FM, FN, CON, DEL, NR, X(1,501), Z(1,501), FREQU(3), DET(3), AIN(4),
1BIN(4), CIN(4), CRAD, DPLA, IDFT, ILU, KLUF, UCO, RCO, RADIUS(501)
CDOUBLE PRECISION F321, F213, FD, AF1, F23, F31, F12, D1F1, D2F2, D3F3, AF2,
1F132, AF3, QD, CC, SR1, FF1, FF2
IF(IDFT-2) 10, 20, 30
10 FREQU(2)=FREQU(1)
FREQU(1)=FREQU(2)+DPLA
DET(2)=DET(1)
NCLUF= 3
GO TO 40
20 DDET = DET(1)* DET(2)
NCLUF = 3
IF(DDET) 50, 50, 60
60 ILU = 2
GO TO 70
50 ILU = 1
70 FREQU(3)=FREQU(2)
DET(3) = DET(2)
FREQU(2) = FREQU(1)
DET(2) = DET(1)
95 GO TO (90,100), ILU
90 FREQU(1)=.5*((FREQU(3)+FREQU(2))-((FREQU(3)-FREQU(2))/(DET(3)-
X DET(2)))*(DET(3)+DET(2)))
GO TO 40
100 FREQU(1)=FREQU(2)+DPLA
GO TO 40
GO TO(110,120), ILU
110 DDET = DET(1)*DET(2)
IF(DDET)130,130,140
130 NCLUF =1
GO TO 150
140 DDET=DET(1)*DET/3)
NCLUF = 2
IF(DDET)150,150,70
150 IF(FREQU(3))165,165,300
165 GO TO(175,185), NCLUF
175 FREQU(3)=FREQU(1)
DET(3)=DET(1)
GO TO 95
185 FREQU(2)=FREQU(1)
DET(2)=DET(1)
GO TO 95
300 DF12=ABS ((FREQU(1)-FREQU(2))/FREQU(1))
DF13=ABS ((FREQU(1)-FREQU(3))/FREQU(1))
DF23=ABS ((FREQU(2)-FREQU(3))/FREQU(2))
IF(DF12-.1F-04)312,312,313
313 IF(DF13-.1F-04)312,312,314
314 IF(DF23-.1F-04)312,312,311
312 GO TO(165,50), ILU
311 C1 = DET(1)
C2 = DET(2)
C3 = DET(3)
F1 = FREQU(1)
F2 = FREQU(2)

```

PDIAR - FFN SOURCE STATEMENT - IFN(S) -

```

F3 = FRFQU(3)                                00000172
F321 = (F3-F21)/F1                            00000173
F132 = (F1-F3)/F2                            00000174
F213 = (F2-F1)/F3                            00000175
FD = F321 + F132 + F213                      00000176
AF1 = (D1*F321 + D2*F132 + D3*F213)/FD        00000177
F23 = F2/F3                                    00000178
F31 = F3/F1                                    00000179
F12 = F1/F2                                    00000180
D1F1 = D1/F1                                    00000181
D2F2 = D2/F2                                    00000182
D3F3 = D3/F3                                    00000183
AF2 = (D1F1*(F23-1.0/F23)+D2F2*(F31-1.0/F31)+D3F3*(F12-1.0/F12))/ 00000184
1FD                                             00000185
AF3 = (D1F1*(1.0/F2-1.0/F3)+D2F2*(1.0/F3-1.0/F1)+D3F3*(1.0/F1-1.0 00000186
1/F2))/FD                                       00000187
QD = AF1/AF3                                   00000188
CC = 0.5*AF2/AF3                             00000189
CCD=CC**2-QD                                   00000190
IF(CCD)312,312,397                            00000191
397 SR1 = SORT (CCD)                          00000192
FF1 = -CC+SR1                                  00000193
FF2 = -CC-SR1                                  00000194
GO TO(160,400),ILU                             00000195
400 ILU = 1                                     00000196
FRFQU(3) = FRFQ                                00000197
DE(3) = DET(2)                                00000198
FREQU(2) = FRFQU(1)                            00000199
DET(2) = DET(1)                                00000200
C                                                00000201
80 IF(FF1-FRFQU(3))500,500,600                00000202
500 FRFQU(1)=FF2                               00000203
GO TO 40                                         00000204
600 IF(FF1-FRFQU(2))700,700,500              00000205
700 FRFQU(1)=FF1                               00000206
GO TO 40                                         00000207
160 GO TO(170,180),NCLUF                      00000208
170 FRFQU(3)=FRFQU(1)                         00000209
DET(3) =DET(1)                                00000210
GO TO 80                                        00000211
180 FREQU(2)=FRFQU(1)                         00000212
DET(2) =DET(1)                                00000213
GO TO 80                                        00000214
120 DDET = DET(1)*DET(2)                      00000215
IF(DDET)190,190,200                           00000216
190 NCLUF = 2                                  00000217
IF(FRFQU(3))50,50,300                         00000218
200 NCLUF = 3                                  00000219
GO TO 70                                         00000220
40 GO TO(210,220,230),NCLUF                  00000221
210 DFRA = ABS (FREQU(1)-FREQU(3))            00000222
GO TO 240                                        00000223
220 DFRA = ABS (FREQU(1)-FREQU(2))            00000224
IF(DFRA-.1F-07)250,250,230                   00000225
230 KLUF = 1                                    00000226
GO TO(260,270,280),NCLUF                     00000227

```

REFDDT

POI AR

- EFN

SOURCE STATEMENT - IFN(S) -

02/22/66

260 FREQU(1) = FREQU(3)
GO TO 280
270 FREQU(1) = FREQU(2)
GO TO 280
230 KIUF = 2
280 CONTINUE
RETURN
END

00000228
00000229
00000230
00000231
00000232
00000233
00000234
00000235

```
SUBROUTINE RKGCON(AIN,BIN,CIN)
DIMENSION AIN(4),BIN(4),CIN(4)
AIN(1)=1./2.
SRT=SQRT(AIN(1))
AIN(2)=1.-SRT
AIN(3)=1.+SRT
AIN(4)=1./6.
BIN(1)=2.
BIN(2)=1.
BIN(3)=1.
BIN(4)=2.
CIN(1)=AIN(1)
CIN(2)=AIN(2)
CIN(3)=AIN(3)
CIN(4)=AIN(1)
RETURN
END
```

SUBROUTINE RESCON(AI,RI)

DIMENSION AI(50),RI(50)

AI(1) = 0.0452984

AI(2) = 0.0822831

AI(3) = 0.1123775

AI(4) = 0.1367632

AI(5) = 0.1564208

AI(6) = 0.1721644

AI(7) = 0.1846699

AI(8) = 0.1944987

AI(9) = 0.2021165

AI(10) = 0.2079104

AI(11) = 0.2122016

AI(12) = 0.2152568

AI(13) = 0.2172976

AI(14) = 0.2185076

AI(15) = 0.2190694

AI(16) = 0.2190195

AI(17) = 0.2185528

AI(18) = 0.2177263

AI(19) = 0.2166120

AI(20) = 0.2152693

AI(21) = 0.2137478

AI(22) = 0.2120877

AI(23) = 0.2103230

AI(24) = 0.2084811

AI(25) = 0.2065846

AI(26) = 0.2046523

AI(27) = 0.2026990

AI(28) = 0.2007374

AI(29) = 0.1987773

AI(30) = 0.1968267

AI(31) = 0.1948921

AI(32) = 0.1929786

AI(33) = 0.1910902

AI(34) = 0.1892299

AI(35) = 0.1873999

AI(36) = 0.1856022

AI(37) = 0.1838379

AI(38) = 0.1821076

AI(39) = 0.1804119

AI(40) = 0.1787508

AI(41) = 0.1771244

AI(42) = 0.1755325

AI(43) = 0.1739746

AI(44) = 0.1724502

AI(45) = 0.1709588

AI(46) = 0.1694997

AI(47) = 0.1680723

AI(48) = 0.1666757

AI(49) = 0.1653093

AI(50) = 0.1639723

RI(1) = .9071009

RI(2) = .8269385

RI(3) = .7575806

ISUM

ACTUALLY
ISUB1

ISUBM+1

ACTUALLY

REDDI

GOOF4

- EFN SOURCE STATEMENT - IFN(S) -

02/09/66

ISUBZERO

BI(4) = .6974022
BI(5) = .6450353
BI(6) = .5993272
BI(7) = .5593055
BI(8) = .5241489
BI(9) = .4931630
BI(10) = .4657596
BI(11) = .4414404
BI(12) = .4197821
BI(13) = .4004249
BI(14) = .3830625
BI(15) = .3674336
BI(16) = .3533150
BI(17) = .3405157
BI(18) = .3288719
BI(19) = .3182432
BI(20) = .3085083
BI(21) = .2995631
BI(22) = .2913173
BI(23) = .2836930
BI(24) = .2766223
BI(25) = .2700464
BI(26) = .2639140
BI(27) = .2581801
BI(28) = .2528055
BI(29) = .2477557
BI(30) = .2430003
BI(31) = .2385126
BI(32) = .2342688
BI(33) = .2302480
BI(34) = .2264314
BI(35) = .2228024
BI(36) = .2193462
BI(37) = .2160494
BI(38) = .2129001
BI(39) = .2098875
BI(40) = .2070019
BI(41) = .2042345
BI(42) = .2015774
BI(43) = .1990232
BI(44) = .1965656
BI(45) = .1941983
BI(46) = .1919160
BI(47) = .1897134
BI(48) = .1875862
BI(49) = .1855300
BI(50) = .1835408
RETURN
END

Unclassified
Security Classification

DOCUMENT CONTROL DATA - R&D

(Security classification of title, body of abstract and indexing annotation must be entered when the overall report is classified)

1. ORIGINATING ACTIVITY (Corporate author) The Franklin Institute Research Laboratories Benjamin Franklin Parkway Philadelphia, Pennsylvania		2a. REPORT SECURITY CLASSIFICATION Unclassified	
		2b. GROUP	
3. REPORT TITLE ON THE EIGENVALUES OF COUETTE FLOW IN A FULLY-FILLED CYLINDRICAL CONTAINER			
4. DESCRIPTIVE NOTES (Type of report and inclusive dates)			
5. AUTHOR(S) (Last name, first name, initial) Reddi, M. M.			
6. REPORT DATE January 1967		7a. TOTAL NO. OF PAGES	7b. NO. OF REFS
8a. CONTRACT OR GRANT NO. DA-30-069-AMC-686(R)		9a. ORIGINATOR'S REPORT NUMBER(S) Technical Report No. F-B2294	
b. PROJECT NO.		9b. OTHER REPORT NO(S) (Any other numbers that may be assigned this report)	
c.			
d.			
10. AVAILABILITY/LIMITATION NOTICES Distribution of this document is unlimited.			
11. SUPPLEMENTARY NOTES		12. SPONSORING MILITARY ACTIVITY U.S. Army Ballistic Research Laboratories Aberdeen Proving Ground, Maryland	
13. ABSTRACT For a stationary flow in a cylindrical container of the Couette type in an outer radial zone, and of zero velocity in an inner radial zone, the normal mode equations are derived. For negative wave numbers in the θ -direction, these equations are found to have a singularity. The eigenvalues are calculated by initial value methods employing the Runge-Kutta-Gill integration procedure. Values of the dependent function and their derivatives at the singularity are calculated by linear extrapolation coupled with continuity requirements. Tables of eigenvalues for various slenderness ratios of the cylinder and various radial nodes are given for θ -wave numbers of -1.			

14. KEY WORDS	LINK A		LINK B		LINK C	
	ROLE	WT	ROLE	WT	ROLE	WT
Eigenvalues, Viscous Fluid Couette Flow Rotating Cylinders						

INSTRUCTIONS

1. ORIGINATING ACTIVITY: Enter the name and address of the contractor, subcontractor, grantee, Department of Defense activity or other organization (*corporate author*) issuing the report.

2a. REPORT SECURITY CLASSIFICATION: Enter the overall security classification of the report. Indicate whether "Restricted Data" is included. Marking is to be in accordance with appropriate security regulations.

2b. GROUP: Automatic downgrading is specified in DoD Directive 5200.10 and Armed Forces Industrial Manual. Enter the group number. Also, when applicable, show that optional markings have been used for Group 3 and Group 4 as authorized.

3. REPORT TITLE: Enter the complete report title in all capital letters. Titles in all cases should be unclassified. If a meaningful title cannot be selected without classification, show title classification in all capitals in parenthesis immediately following the title.

4. DESCRIPTIVE NOTES: If appropriate, enter the type of report, e.g., interim, progress, summary, annual, or final. Give the inclusive dates when a specific reporting period is covered.

5. AUTHOR(S): Enter the name(s) of author(s) as shown on or in the report. Enter last name, first name, middle initial. If military, show rank and branch of service. The name of the principal author is an absolute minimum requirement.

6. REPORT DATE: Enter the date of the report as day, month, year; or month, year. If more than one date appears on the report, use date of publication.

7a. TOTAL NUMBER OF PAGES: The total page count should follow normal pagination procedures, i.e., enter the number of pages containing information.

7b. NUMBER OF REFERENCES: Enter the total number of references cited in the report.

8a. CONTRACT OR GRANT NUMBER: If appropriate, enter the applicable number of the contract or grant under which the report was written.

8b, 8c, & 8d. PROJECT NUMBER: Enter the appropriate military department identification, such as project number, subproject number, system numbers, task number, etc.

9a. ORIGINATOR'S REPORT NUMBER(S): Enter the official report number by which the document will be identified and controlled by the originating activity. This number must be unique to this report.

9b. OTHER REPORT NUMBER(S): If the report has been assigned any other report numbers (*either by the originator or by the sponsor*), also enter this number(s).

10. AVAILABILITY/LIMITATION NOTICES: Enter any limitations on further dissemination of the report, other than those imposed by security classification, using standard statements such as:

(1) "Qualified requesters may obtain copies of this report from DDC."

(2) "Foreign announcement and dissemination of this report by DDC is not authorized."

(3) "U. S. Government agencies may obtain copies of this report directly from DDC. Other qualified DDC users shall request through _____."

(4) "U. S. military agencies may obtain copies of this report directly from DDC. Other qualified users shall request through _____."

(5) "All distribution of this report is controlled. Qualified DDC users shall request through _____."

If the report has been furnished to the Office of Technical Services, Department of Commerce, for sale to the public, indicate this fact and enter the price, if known.

11. SUPPLEMENTARY NOTES: Use for additional explanatory notes.

12. SPONSORING MILITARY ACTIVITY: Enter the name of the departmental project office or laboratory sponsoring (*paying for*) the research and development. Include address.

13. ABSTRACT: Enter an abstract giving a brief and factual summary of the document indicative of the report, even though it may also appear elsewhere in the body of the technical report. If additional space is required, a continuation sheet shall be attached.

It is highly desirable that the abstract of classified reports be unclassified. Each paragraph of the abstract shall end with an indication of the military security classification of the information in the paragraph, represented as (TS), (S), (C), or (U).

There is no limitation on the length of the abstract. However, the suggested length is from 150 to 225 words.

14. KEY WORDS: Key words are technically meaningful terms or short phrases that characterize a report and may be used as index entries for cataloging the report. Key words must be selected so that no security classification is required. Identifiers, such as equipment model designation, trade name, military project code name, geographic location, may be used as key words but will be followed by an indication of technical context. The assignment of links, rules, and weights is optional.